

# The Role of Transverse Flow in Co-Injection Resin Transfer Molding

by Bruce K. Fink, Emanuele F. Gillio, Suresh G. Advani, and John W. Gillespie Jr.

ARL-TR-2135 December 1999

Approved for public release; distribution is unlimited.

20000118 053

The findings in this report are not to be construed as an official Department of the Army position unless so designated by other authorized documents.

Citation of manufacturer's or trade names does not constitute an official endorsement or approval of the use thereof.

Destroy this report when it is no longer needed. Do not return it to the originator.

# **Army Research Laboratory**

Aberdeen Proving Ground, MD 21005-5069

**ARL-TR-2135** 

December 1999

# The Role of Transverse Flow in Co-Injection Resin Transfer Molding

Bruce K. Fink Weapons and Materials Research Directorate

Emanuele F. Gillio, Suresh G. Advani, and John W. Gillespie Jr. University of Delaware

Approved for public release; distribution is unlimited.

#### **Abstract**

A co-injection resin transfer molding (CIRTM) process has been developed at the U.S. Army Research Laboratory (ARL) in collaboration with the University of Delaware. It enables two or more resins to be simultaneously injected into a mold filled with a stationary fiber preform. This process allows for the manufacturing of cocured multilayer multiresin structures in a single processing step. A separation layer is used to provide resin compatibility during cure and to control resin mixing. In this study, scaling issues relating the role of transverse permeability in resin mixing are investigated. This report presents two different approaches taken to understand the causes of transverse flow and to quantify the amount of transverse flow. The first approach is a one-dimensional (1-D) model, which explains the important parameters that govern the flow in CIRTM. The second approach is based on an existing finite element code, which is modified to allow for the injection of multiple resins. The total amount of transverse flow was quantified using the finite element code. This research shows that he CIRTM process requires a totally impermeable separation layer if CIRTM is used to manufacture large parts and/or if the resins injected have significantly different viscosities.

# Acknowledgments

The authors would like to thank Murat Sozer for his help in developing the flow modeling input script.

INTENTIONALLY LEFT BLANK.

# **Table of Contents**

		Page
	Acknowledgments	iii
	List of Figures	vii
1.	Background and Motivation	1
2.	One-Dimensional (1-D) Model	2
3.	Finite Element Model	7
4.	Transverse Permeability	9
5.	Part Length Considerations	11
6.	Conclusions	12
7.	References	15
	Distribution List	17
	Report Documentation Page	27

INTENTIONALLY LEFT BLANK.

# **List of Figures**

<u>Figure</u>		Page
1.	Cross Section of Integral Armor	. 1
2.	Schematic of the 1-D Flow Model in CIRTM	2
3.	Effect of the Transverse Permeability on Distance Between Flow Fronts in Top and Bottom Layers	6
4.	Total Transverse Flow as Percentage of Half of Mold Volume for Different Values of Transverse Permeability	· 7
5.	Schematic of Finite Element Mesh Used to Model Co-Injection Process	9
6.	Total Transverse Flow for Different Transverse Permeabilities	10
7.	Percentage of Cross Flow With Respect to Half the Mold Volume as a Function of the Ratio of In-Plane to Transverse Permeability	11
8.	Effect of Part Length on Total Percentage of Transverse Flow	12
9.	Percentage of Transverse Flow as a Function of Part Length	13

INTENTIONALLY LEFT BLANK

## 1. Background and Motivation

Several composites application require multiple layers each serving a specific task while being integrated together in a single structure. For example, lightweight composite armor (Figure 1) incorporates different layers of different materials, which serve as ballistic protection, structural reinforcements, fire smoke and toxicity barriers, and signature management, while being integrated together into a single structure. Currently, the manufacturing of these kind of multilayer structures takes place in multiple steps in which each layer is manufactured individually and then bonded to the others. This process is not cost effective, and the secondary bonding procedures can introduce a number of defects into the final part. Pike, McArthur, and Schade [1] have shown that vacuum-assisted resin transfer molding (VARTM) processes can introduce significant cost savings, but they have been limited to a single resin system.

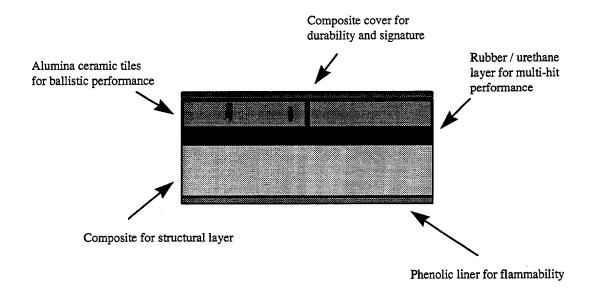


Figure 1. Cross Section of Integral Armor.

Co-injection resin transfer molding (CIRTM) is a new process developed by the U.S. Army Research Laboratory (ARL) and the University of Delaware that enables the manufacturing of multilayer hybrid structures in a single processing step [2]. CIRTM takes advantage of the unique properties and the cost effectiveness of existing liquid molding processes and improves

them by reducing the number of steps necessary to manufacture multilayer structures. Gillio [3] and Gillio et al. [4] give a complete overview of the co-injection process.

The applications of this process are not limited to integral armor. A number of applications exist for multilayer structures in which a thick structural layer of vinyl ester or polyester is combined with a thin layer of phenolic resin for fire, smoke, and toxicity protection. These applications include navy ship decks, containers to transport goods, rail cars, and anywhere composites want to be applied where flammability of materials is of concern.

# 2. One-Dimensional (1-D) Model

A simplified schematic of the co-injection setup is shown in Figure 2. In the majority of cases, the flow of a polymer inside a mold filled with a stationary fiber bed is modeled using Darcy's law [5]. The macroscopic velocity, u, is given by

$$u = \frac{\kappa}{\mu} \frac{dP}{dx},\tag{1}$$

where  $\kappa$  is the permeability of the fiber preform, and  $\mu$  is the viscosity of the resin. In co-injection, the top and the bottom preforms can have different permeabilities and the resins injected can have different viscosities leading to different resin velocities between the top and the bottom halves of the mold. The injection is assumed to be at constant pressure since this is the case in the majority of VARTM processes.

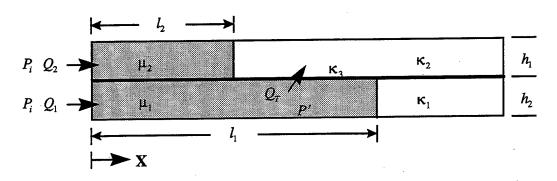


Figure 2. Schematic of 1-D Flow Model in CIRTM.

In such a case, the velocities of the two flow fronts is given by

$$u_1 = \frac{\kappa_1}{\mu_1} \frac{P_i}{l_1} \tag{2}$$

and

$$u_2 = \frac{\kappa_2}{\mu_2} \frac{P_i}{l_2}.\tag{3}$$

It follows that, if the ratios of permeability to viscosity are the same in both layers, the flow velocities will be the same. In most applications, this is not the case and one of the flow fronts will move ahead of the other.

As the flow moves through the mold, a pressure profile develops in both the top and the bottom halves of the mold. In the case where the cross flow between the two layers is small, these pressure profiles can be assumed to be linear. In this model, both pressure profiles are assumed to be linear. If the ratio of permeability to viscosity is different between the two layers,  $l_1$  and  $l_2$  will be different and a pressure gradient will form between the top and the bottom halves of the mold. This pressure gradient will drive the transverse flow. The transverse flow will be given by

$$u_3 = \frac{2\kappa_3}{\mu_1} \frac{P'}{(h_1 + h_2)},\tag{4}$$

where  $\kappa_3$  is the transverse permeability of the middle layer,  $(h_1 + h_2)/2$  is the distance between the midpoints of the two layers, and P' is the average pressure difference between the two layers. Since both pressure profiles are assumed to be linear, and recalling that the pressure is zero in front of the fluid front (i.e., where the preform is not impregnated), P' is given by

$$P' = \frac{P_i}{l_1} \frac{(l_1 - l_2)}{2}. (5)$$

Here, the assumption has been made that the ratio of permeability to viscosity of layer one is greater than that of layer two. Substituting equation (5) into equation (4), an equation for the cross flow is obtained as a function of the inlet pressure and the position of the flow fronts:

$$u_3 = \frac{\kappa_3}{\mu_1} \frac{P_i(l_1 - l_2)}{l_1(h_1 + h_2)}.$$
 (6)

In equation (4) and (6), the viscosity of fluid one was used, assuming that the ratio of permeability to viscosity of layer one is greater than that of layer two, which is the case in Figure 2. Equation (6) can be verified in the extreme cases: if the transverse permeability is zero, the transverse flow is zero; if the two flow fronts are at the same location (i.e.,  $l_1 = l_2$ ), there is no pressure gradient and the transverse flow is zero. A mass balance yields

$$Q_1 - Q_T = Q_2 + Q_T, (7)$$

where  $Q_1$  and  $Q_2$  are the flow rates through injection gates one and two, respectively, and  $Q_T$  is the flow rate through the separation layer. Assuming unit width of the part, the flow rates are given by

$$Q_1 = \frac{\kappa_1}{\mu_1} \frac{P_i}{l_1} h_1, \tag{8}$$

$$Q_2 = \frac{\kappa_2}{\mu_2} \frac{P_i}{l_2} h_2, \tag{9}$$

and

$$Q_T = \frac{\kappa_3}{\mu_1} \frac{P_i (l_1 - l_2)^2}{l_i (h_1 + h_2)}.$$
 (10)

Knowing the flow rates, it is possible to compute the flow progression in the x direction based on a given time step, dt. The positions of the two flow fronts can be obtained using the following expressions:

$$l_1 = l_1 + \frac{(Q_1 - Q_T) \cdot dt}{h_1 \cdot V_{f1}}, \tag{11}$$

and

$$l_2 = l_2 + \frac{(Q_2 + Q_T) \cdot dt}{h_2 \cdot V_{f2}},\tag{12}$$

where  $V_{f1}$  and  $V_{f2}$  are the respective fiber volume fractions of each preform.

Clearly, from equations (2) and (3), the flow in the layer with the highest ratio of permeability to viscosity will be ahead. However, the distance between flow fronts will not increase continuously but, rather, it will slowly approach a steady-state scenario. The exact time, or position, at which this occurs depends on the total amount of transverse flow, which depends on a number of other factors such as the transverse permeability and the pressure gradient formed in the transverse direction. Figure 3 shows a plot of the distance between flow fronts vs. the flow front positions for a variety of different transverse permeabilities. All of the cases eventually reach steady state. It should be noted that the nondimensional ratio R was used in Figure 3 because it emphasizes the various parameters that affect the resin flow. However, in this graph, only the transverse permeability was changed.

This behavior can also be observed in the equations. Substituting equations (8), (9), and (10) into equation (7) and assuming the two layers have the same thickness, h, and the fluids the same viscosity, the distance between the two flow fronts, normalized by the thickness, is given by

$$\frac{l_1 - l_2}{h} = -\frac{\kappa_2 h}{2l_2 \kappa_3} + \frac{1}{2} \sqrt{\left(\frac{\kappa_2 h}{l_2 \kappa_3}\right)^2 + 4\frac{\kappa_1 - \kappa_2}{\kappa_3}}.$$
 (13)

If the permeabilities of the two preforms are the same, the distance between the two flow fronts reduces to zero, as would be expected. Additionally, as  $l_2$  becomes large, the distance between the two flow fronts becomes constant. Taking the limit of equation (13) as  $l_2$  approaches infinity yields

$$\frac{l_1 - l_2}{h} = \sqrt{\frac{\kappa_1 - \kappa_2}{\kappa_3}}. (14)$$

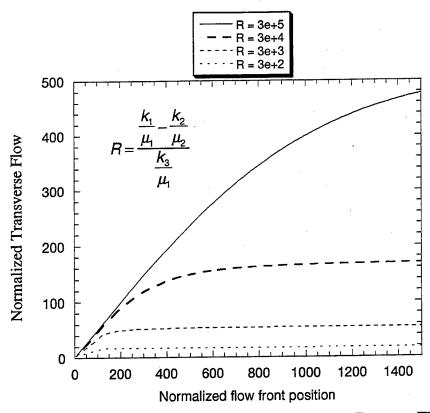


Figure 3. Effect of Transverse Permeability on Distance Between Flow Fronts in Top and Bottom Layers. All Variables Are Normalized With Respect to the Thickness of One Layer.

This equation was originally introduced by Bruschke [6].

The term under the square root on the right-hand side of equation (14) is exactly the same as the ratio R that was used in Figure 3, except that the viscosities were assumed to be the same and therefore they canceled out. The second important result that can be obtained from a simple 1-D model is the total amount of transverse flow. Figure 4 shows the effect of transverse permeability on the total amount of transverse flow. In this case, the permeabilities of the two layers were assumed to be the same, while the viscosity of one resin was four times that of the other. Under these conditions, different transverse permeabilities were used in an attempt to control the transverse flow. The trend observed is that the transverse flow increases rapidly in the initial stages of mold filling and, subsequently, when the distance between the flow fronts becomes constant, the total transverse flow does not increase significantly.

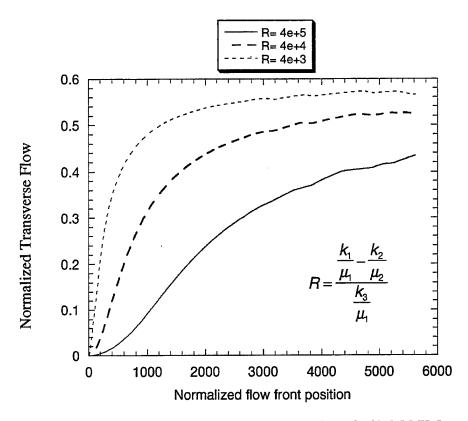


Figure 4. Total Transverse Flow as Percentage of Half of Mold Volume for Different Values of Transverse Permeability. The Flow Front Position Is Normalized With Respect to the Thickness of the Layer.

#### 3. Finite Element Model

Liquid Injection Molding Simulation (LIMS) is a finite element code developed at the University of Delaware [7–12]. It is a two-dimensional (2-D) flow modeling package that can deal with complex geometries and multiple injection gates and vent locations. In traditional VARTM processes, the thickness is much smaller than the other two dimensions. The flow in the thickness direction can therefore be neglected, and a 2-D model can be used. However, this is not the case in CIRTM because the flow in the thickness direction plays an important role in determining the amount of mixing that takes place between the two resins.

In LIMS, the user creates the geometry of the part using a commercial finite element code. Once the geometry is created, it is converted into a file that is readable for LIMS. The user must then enter all of the material properties (i.e., resin viscosity, preform permeability, etc.) and the processing parameters (i.e., injection gates location, inlet pressure or flow rates, etc.). LIMS can then process all of this information and reproduce the pressure field in the mold, the filling history, etc. To simulate the co-injection process, the program was slightly modified. Only one type of resin can be injected into the mold using LIMS, but multiple injection gates are possible. Since the flow patterns are governed by the ratios of viscosity to permeability, the permeabilities in the top and the bottom layers were modified to account for the injection of resins with different viscosities.

The geometry used is shown in Figure 5, where the flow was modeled in the one and the three direction to be able to account for the transverse flow. LIMS is based on a control volume finite element approach [7]. It associates a control volume with every node in the finite element mesh (Figure 5) and, as the flow front moves, it assigns a fill factor to each node. If the control volume of a specific node is empty, the fill factor will be zero; if it is completely full, the fill factor will be one; and if it is partially full, the fill factor will reflect the percentage of the control volume that is filled. A script was written that calculates the volume of every node; it then uses the fill factors at each time step to determine how much fluid has entered the bottom half and the top half of the mold (Figure 5). These values are then compared to the amount of fluid injected through the two injection gates. The difference between fluid injected in one layer and volume filled in the same layer is computed, and this will be the transverse flow for this time step from mass conservation. Additionally, at every time step, the script compares the total fluid injected through both gates to the total volume filled in the two halves of the mold; the ratio of these two values should be one at every step. Finally, the accuracy of the results was checked by mesh refinement.

A parametric study was conducted on two parameters that the 1-D model had shown to be fundamental to understanding the transverse flow. These parameters were the transverse permeability of the separation layer and the total length of the part. The 1-D model emphasized

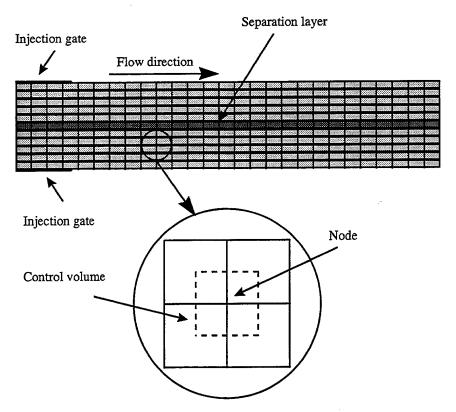


Figure 5. Schematic of Finite Element Mesh Used to Model Co-Injection Process.

other parameters, particularly the difference between the ratio of permeability to viscosity of the two layers. However, these values were maintained the same because, in a real world application, once the materials are selected, their properties vary only slightly.

# 4. Transverse Permeability

To better understand the role and importance of this parameter, a parametric study was conducted. The minimum transverse permeability was selected so that the total transverse flow would be less then 1% of half of the mold volume. The maximum transverse permeability was selected to be one order of magnitude greater then the in plane permeability. Figure 6 shows a comparison between the different transverse flows obtained. As expected, the transverse flow increases with decreasing transverse permeability (i.e., increasing R ratio). As before, the R ratio was used to emphasize all parameters that can effect the transverse flow. However,

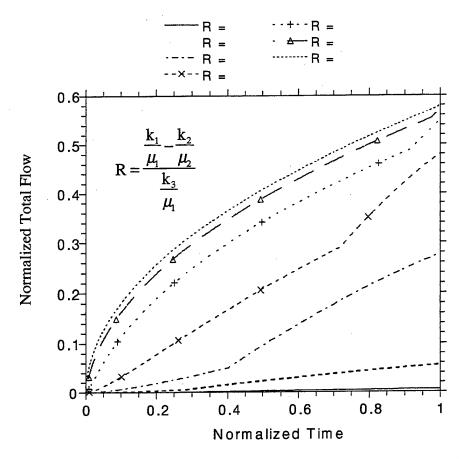


Figure 6. Total Transverse Flow for Different Transverse Permeabilities. Transverse Flow Is Normalized With Respect to Half of the Mold Volume. Time Is Normalized With Respect to Total Filling Time.

permeabilities and viscosities of the two layers were maintained constant. Additionally, the filling time decreases with increasing transverse flow. This happens because the maximum fill time is the time it would take the higher viscosity resin to fill its half of the mold without any cross flow. The transverse flow can be thought of as an additional inlet, so, as the transverse flow increases, it will take less time to fill the mold. One final behavior that should be observed is that the total transverse flow does not increase linearly with decreasing permeability. Figure 7 shows the total transverse flow as a function of transverse permeability. The transverse flow increases rapidly when the transverse permeability decreases from 6 orders of magnitude greater then the in plane permeability to 4 orders of magnitude. After this point, it slowly levels off.

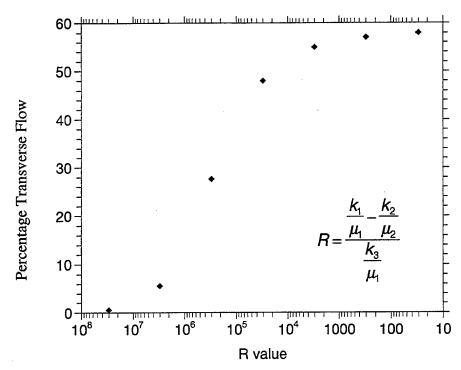


Figure 7. Percentage of Cross Flow With Respect to Half the Mold Volume as a Function of the Ratio of In-Plane to Transverse Permeability.

The results from this parametric study show that it is extremely difficult to control the transverse flow. The only case in which it was possible to limit the transverse flow to less then 1% was by using an extremely small transverse permeability. In practice, it is very difficult to find a separation layer with a transverse permeability that is approximately 7 orders of magnitude smaller than the in-plane permeability of a reinforcing preform. Consequently, this parametric study supports our experimental work showing that an impermeable layer is required.

### 5. Part Length Considerations

The 1-D model also emphasized the effect of the part length on the transverse flow. Figure 4 showed that the amount of transverse flow eventually levels off, but this only happens after considerably long distances. The parametric study was conducted by varying the length of the part while maintaining all other parameters constant. Figure 8 shows the total transverse flow as a function of time. Time has been normalized with respect to the total fill time of each mold.

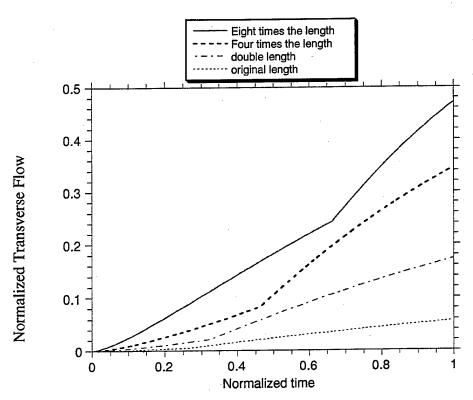


Figure 8. Effect of Part Length on Total Percentage of Transverse Flow. Transverse Flow Is Normalized With Respect to Half the Mold Size. Time Is Normalized With Respect to Total Filling Time.

The percentage cross flow is computed with respect to half of the mold volume. Clearly, the amount of cross flow increases with increasing mold size. This is also reflected in Figure 9, which suggests that the transverse flow does not increase linearly but, rather, eventually levels off. This behavior is expected because it was also observed in the 1-D model.

#### 6. Conclusions

The goal of understanding the parameters that govern the flow in CIRTM was met using the 1-D model. In CIRTM, as the resins fill the two layers in the mold, two pressure profiles develop. These profiles are determined by the ratio of the permeability of the preform to the viscosity of the resin. Since, in the majority of cases the two ratios are different, two different pressure profiles develop in the top and bottom layer. When this occurs, a pressure gradient

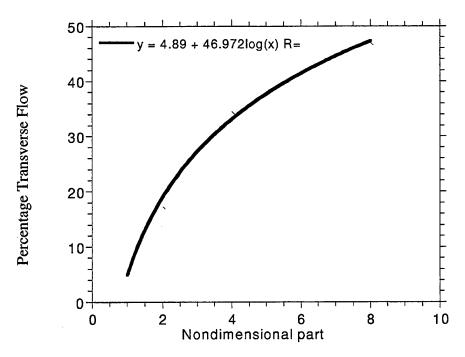


Figure 9. Percentage of Transverse Flow as a Function of Part Length. Log Curve Fit Is Shown.

forms in the transverse direction and drives the transverse flow. The model also showed that the amount of transverse flow increases rapidly until it reaches a steady-state point. All of the results from this model qualitatively matched the ones from the finite element model.

The finite element model allowed measuring the amount of transverse flow and determining the importance of the transverse permeability and the effect of part length on the transverse flow. The model also relaxed the assumption of linear pressure profile in each half of the mold. It showed that a transverse permeability 7 orders of magnitude smaller then the in-plane permeability is needed to reduce the transverse flow to less then 1% of half the mold size. In practice, it is almost impossible to find a preform material with this property. Additionally, the finite element code proved that increasing the part length and, therefore, the distance that the resin must travel, increases the amount of transverse flow. As in the 1-D model, the amount of transverse flow seems to eventually reach a steady-state situation but only after a significant amount of mixing has occurred.

Overall, the results from the two models agree qualitatively and also agree with preliminary experiments conducted in the lab. In order to use the co-injection process for large structures, it is necessary to use a completely impermeable separation layer. This will completely eliminate transverse flow. For smaller scale applications, the results presented in this study provide useful guidance for selection of resin and preforms to minimize transverse flow.

#### 7. References

- 1. Pike, T., M. McArthur, and D. Schade. "Vacuum Assisted Resin Transfer Molding of a Layered Structural Laminate for Application on Ground Combat Vehicles." *Proceedings of the 28th International SAMPE Technical Conference*, pp. 374–380, Seattle, WA, 4–7 November 1996.
- 2. Fink, B. K., J. W. Gillespie, Jr., E. F. Gillio, and K. R. Bernetich. "One-Step Resin Transfer Molding of Multi-Functional Composites Consisting of Multiple Resins." Patent application filed with the U.S. Patent Office, 22 October 1997.
- 3. Gillio, E. F. "Co-Injection Resin Transfer Molding of Hybrid Composites." MS Thesis, University of Delaware, 1998.
- 4. Gillio, E. F., J. W. Gillespie, Jr., R. F. Eduljee, S. G. Advani, K. R. Bernetich, and B. K. Fink. "Manufacturing of Composites With the Co-injection Process." *Proceedings of the Thirty-Eighth AIAA Structures, Structural Dynamics and Materials Conference*, Kissimmee, FL, 7–10 April 1997.
- 5. Darcy, H. Les Fontaines Publiques de la Ville de Dijon. Dalmont, Paris, 1856.
- 6. Bruschke, M. V. "A Predictive Model for Permeability and Non-Isothermal Flow of Viscous and Shear-Thinning Fluids in Anisotropic Fibrous Media." Ph.D. Thesis, University of Delaware, 1992.
- 7. Bruschke, M. V., and S. G. Advani. "A Finite Element/Control Volume Approach to Mold Filling in Anisotropic Porous Media." *Polymer Composites*, vol. 11, no. 6, pp. 398–405, 1990.
- 8. Liu, B., and S. G. Advani. "Operator Splitting Scheme for 3-D Temperature Solution Based on 2-D Flow Approximation." *Computational Mechanics*, vol. 38, pp. 74–82, 1995.
- 9. Liu, B., S. Bickerton, and S. G. Advani. "Modeling and Simulation of RTM: Gate Control, Venting and Dry Spot Prediction." *Composites Manufacturing*, vol. 27, no. 2, pp. 135–141, 1995.
- 10. Maier, R. S., T. F. Rohaly, S. G. Advani, and K. D. Fickie. "A Fast Numerical Method for Isothermal Resin Transfer Mold Filling." *International Journal of Numerical Methods in Engineering*, vol. 39, pp. 1405-1422, 1996.
- 11. Bruschke, M. V., and S. G. Advani. "A Numerical Approach to Model Non-Isothermal, Viscous Flow With Free Surfaces Through Fibrous Media." *International Journal of Numerical Methods in Fluids*, vol. 19, pp. 575–603, 1994.

12. Bruschke, M. V., and S. G. Advani. "RTM: Filling Simulation of Complex Three-Dimensional Shell-Like Structures." *SAMPE Quarterly*, vol. 23, no. 1, pp. 2–11, 1991.

# NO. OF COPIES ORGANIZATION

- 2 DEFENSE TECHNICAL INFORMATION CENTER DTIC DDA 8725 JOHN J KINGMAN RD STE 0944 FT BELVOIR VA 22060-6218
- 1 HQDA
  DAMO FDQ
  D SCHMIDT
  400 ARMY PENTAGON
  WASHINGTON DC 20310-0460
- 1 OSD
  OUSD(A&T)/ODDDR&E(R)
  R J TREW
  THE PENTAGON
  WASHINGTON DC 20301-7100
- 1 DPTY CG FOR RDA
  US ARMY MATERIEL CMD
  AMCRDA
  5001 EISENHOWER AVE
  ALEXANDRIA VA 22333-0001
- 1 INST FOR ADVNCD TCHNLGY THE UNIV OF TEXAS AT AUSTIN PO BOX 202797 AUSTIN TX 78720-2797
- 1 DARPA
  B KASPAR
  3701 N FAIRFAX DR
  ARLINGTON VA 22203-1714
- 1 NAVAL SURFACE WARFARE CTR CODE B07 J PENNELLA 17320 DAHLGREN RD BLDG 1470 RM 1101 DAHLGREN VA 22448-5100
- 1 US MILITARY ACADEMY
  MATH SCI CTR OF EXCELLENCE
  DEPT OF MATHEMATICAL SCI
  MADN MATH
  THAYER HALL
  WEST POINT NY 10996-1786

# NO. OF COPIES ORGANIZATION

- 1 DIRECTOR
  US ARMY RESEARCH LAB
  AMSRL DD
  J J ROCCHIO
  2800 POWDER MILL RD
  ADELPHI MD 20783-1197
- 1 DIRECTOR
  US ARMY RESEARCH LAB
  AMSRL CS AS (RECORDS MGMT)
  2800 POWDER MILL RD
  ADELPHI MD 20783-1145
- 3 DIRECTOR
  US ARMY RESEARCH LAB
  AMSRL CI LL
  2800 POWDER MILL RD
  ADELPHI MD 20783-1145

#### ABERDEEN PROVING GROUND

4 DIR USARL AMSRL CI LP (BLDG 305)

NO. OF COPIES	ORGANIZATION	NO OF. COPIES	ORGANIZATION
1	DIRECTOR USARL AMSRL CP CA D SNIDER 2800 POWDER MILL RD ADELPHI MD 20783	5	COMMANDER USA ARDEC AMSTA AR CCH S MUSALLI R CARR M LUCIANO
1	COMMANDER USA ARDEC AMSTA AR FSE T GORA PICATINNY ARSENAL NJ	4	T LOUCEIRO PICATINNY ARSENAL NJ 07806-5000 COMMANDER
3	07806-5000 COMMANDER	. 4	USA ARDEC AMSTA AR (2CPS)
J	USA ARDEC AMSTA AR TD PICATINY ARSENAL NJ 078806-5000		E FENNEL (2 CPS) PICATINNY ARSENAL NJ 07806-5000
5	COMMANDER USA TACOM AMSTA JSK S GOODMAN J FLORENCE	1	COMMANDER USA ARDEC AMSTA AR CCH P J LUTZ PICATINNY ARSENAL NJ 07806-5000
	AMSTA TR D B RAJU L HINOJOSA D OSTBERG WARREN MI 48397-5000	1	COMMANDER USA ARDEC AMSTA AR FSF T C LIVECCHIA PICATINNY ARSENAL NJ 07806-5000
5	PM SADARM SFAE GCSS SD COL B ELLIS M DEVINE W DEMASSI J PRITCHARD	1	COMMANDER USA ARDEC AMSTA AR QAC T/C C PATEL PICATINNY ARSENAL NJ 07806-5000
	S HROWNAK PICATINNY ARSENAL NJ 07806-5000	2	COMMANDER USA ARDEC AMSTA AR M D DEMELLA
1	COMMANDER USA ARDEC F MCLAUGHLIN PICATINNY ARSENAL NJ 07806-5000		F DIORIO PICATINNY ARSENAL NJ 07806-5000

NO. OF COPIES	ORGANIZATION	NO OF. COPIES	ORGANIZATION
3	COMMANDER USA ARDEC AMSTA AR FSA A WARNASH	1	COMMANDER USA BELVOIR RD&E CTR STRBE JBC FT BELVOIR VA 22060-5606
	B MACHAK M CHIEFA PICATINNY ARSENAL NJ 07806-5000	2	COMMANDER USA ARDEC AMSTA AR FSB G M SCHIKSNIS
1	COMMANDER SMCWV QAE Q B VANINA BLDG 44 WATERVLIET ARSENAL		D CARLUCCI PICATINNY ARSENAL NJ 07806-5000
1	WATERVLIET NY 12189-4050 COMMANDER	1	US ARMY COLD REGIONS RESEARCH & ENGINEERING CTR P DUTTA
	SMCWV SPM T MCCLOSKEY BLDG 253 WATERVLIET ARSENAL		72 LYME RD HANVOVER NH 03755
8	WATERVLIET NY 12189-4050  DIRECTORECTOR BENET LABORATORIES	1	DIRECTOR USARL AMSRL WT L D WOODBURY 2800 POWDER MILL RD
	AMSTA AR CCB J KEANE J BATTAGLIA	1	ADELPHI MD 20783-1145 COMMANDER
	J VASILAKIS G FFIAR V MONTVORI G DANDREA R HASENBEIN		USA MICOM AMSMI RD W MCCORKLE REDSTONE ARSENAL AL 35898-5247
	AMSTA AR CCB R S SOPOK WATERVLIET NY 12189-4050	1	COMMANDER USA MICOM AMSMI RD ST P DOYLE REDSTONE ARSENAL AL
1 .	COMMANDER SMCWV QA QS K INSCO WATERVLIET NY 12189-4050	1	35898-5247 COMMANDER
1	COMMANDER PRODUCTION BASE MODERN ACTY USA ARDEC		USA MICOM AMSMI RD ST CN T VANDIVER REDSTONE ARSENAL AL 35898-5247
	AMSMC PBM K PICATINNY ARSENAL NJ 07806-5000	3	US ARMY RESEARCH OFFICE A CROWSON K LOGAN J CHANDRA PO BOX 12211 RESEARCH TRIANGLE PARK NC 27709-2211
			-

NO. OF COPIES	<u>ORGANIZATION</u>	NO OF. COPIES	<u>ORGANIZATION</u>
3	US ARMY RESEARCH OFFICE ENGINEERING SCIENCES DIV R SINGLETON G ANDERSON K IYER PO BOX 12211	2	COMMANDER DARPA S WAX 2701 N FAIRFAX DR ARLINGTON VA 22203-1714
	RESEARCH TRIANGLE PARK NC 27709-2211	6	COMMANDER WRIGHT PATTERSON AFB WL FIV
5	PM TMAS SFAE GSSC TMA COL PAWLICKI K KIMKER E KOPACZ R ROESER B DORCY PICATINNY ARSENAL NJ 07806-5000		A MAYER WL MLBM S DONALDSON T BENSON-TOLLE C BROWNING J MCCOY F ABRAMS 2941 P ST STE 1 DAYTON OH 45433
1	PM TMAS SFAE GSSC TMA SMD R KOWALSKI PICATINNY ARSENAL NJ 07806-5000	2	NAVAL SURFACE WARFARE CTR DAHLGREN DIV CODE G06 R HUBBARD CODE G 33 C DAHLGREN VA 22448 NAVAL RESEARCH LAB
3	PEO FIELD ARTILLERY SYSTEMS SFAE FAS PM H GOLDMAN		I WOLOCK CODE 6383 WASHINGTON DC 20375-5000
	T MCWILLIAMS T LINDSAY PICATINNY ARSENAL NJ 07806-5000	1	OFFICE OF NAVAL RESEARCH MECH DIV Y RAJAPAKSE CODE 1132SM ARLINGTON VA 22271
2	PM CRUSADER G DELCOCO J SHIELDS PICATINNY ARSENAL NJ 07806-5000	1	NAVAL SURFACE WARFARE CTR CRANE DIV M JOHNSON CODE 20H4 LOUISVILLE KY 40214-5245
3	NASA LANGLEY RESEARCH CTR MS 266 AMSRL VS W ELBER	1	DAVID TAYLOR RESEARCH CTR SHIP STRUCTURES & PROTECTION DEPT J CORRADO CODE 1702 BETHESDA MD 20084
	F BARTLETT JR C DAVILA HAMPTON VA 23681-0001	2	DAVID TAYLOR RESEARCH CTR R ROCKWELL W PHYILLAIER BETHESDA MD 20054-5000

NO. OF COPIES	ORGANIZATION	NO OF. COPIES	ORGANIZATION
1	DEFENSE NUCLEAR AGENCY INNOVATIVE CONCEPTS DIV R ROHR 6801 TELEGRAPH RD ALEXANDRIA VA 22310-3398	2	DIRECTOR LLNL F ADDESSIO MS B216 J REPPA MS F668 PO BOX 1633 LOS ALAMOS NM 87545
1	EXPEDITIONARY WARFARE DIV N85 F SHOUP 2000 NAVY PENTAGON WASHINGTON DC 20350-2000	3	UNITED DEFENSE LP 4800 EAST RIVER DR P JANKE MS170 T GIOVANETTI MS236
1	OFFICE OF NAVAL RESEARCH D SIEGEL 351 800 N QUINCY ST	4	B VAN WYK MS 389 MINNEAPOLIS MN 55421-1498 DIRECTOR
7	ARLINGTON VA 22217-5660  NAVAL SURFACE WARFARE CTR J H FRANCIS CODE G30 D WILSON CODE G32 R D COOPER CODE G32 E ROWE CODE G33 T DURAN CODE G33 L DE SIMONE CODE G33 DAHLGREN VA 22448	4	SANDIA NATIONAL LAB APPLIED MECHANICS DEPT DIV 8241 W KAWAHARA K PERANO D DAWSON P NIELAN PO BOX 969 LIVERMORE CA 94550-0096
1	COMMANDER NAVAL SEA SYSTEM CMD P LIESE 2351 JEFFERSON DAVIS HIGHWAY ARLINGTON VA 22242-5160	1	BATTALLE C R HARGREAVES 505 KNIG AVE COLUMBUS OH 43201-2681 PACIFIC NORTHWEST LAB
1	NAVAL SURFACE WARFARE CTR M E LACY CODE B02 17320 DAHLGREN RD DAHLGREN VA 22448	_	M SMITH PO BOX 999 RICHLAND WA 99352
1	NAVAL WARFARE SURFACE CTR TECH LIBRARY CODE 323 17320 DAHLGREN RD DAHLGREN VA 22448	1	LLNL M MURPHY PO BOX 808 L 282 LIVERMORE CA 94550
4	DIR LLNL R CHRISTENSEN S DETERESA F MAGMESS M FINGER PO BOX 808 LIVERMORE CA 94550	10	UNIV OF DELAWARE CTR FOR OCMPOSITE MATERIALS J GILLESPIE 201 SPENCER LAB NEWARK DE 19716

NO. OF COPIES	ORGANIZATION	NO OF. COPIES	ORGANIZATION
2	THE U OF TEXAS AT AUSTIN CTR ELECTROMECHANICS A WALLIS	1	NOESIS INC 1110 N GLEBE RD STE 250 ARLINGTON VA 22201-4795
	J KITZMILLER 10100 BURNET RD AUSTIN TX 78758-4497	1	ARROW TECH ASSO 1233 SHELBURNE RD STE D 8 SOUTH BURLINGTON VT 05403-7700
1	AAI CORPORATION T G STASTNY PO BOX 126	5	GEN CORP AEROJET
. 1	HUNT VALLEY MD 21030-0126 SAIC		D PILLASCH T COULTER C FLYNN
•	D DAKIN 2200 POWELL ST STE 1090 EMERYVILLE CA 94608		D RUBAREZUL M GREINER 1100 WEST HOLLYVALE ST AZUSA CA 91702-0296
1	SAIC M PALMER 2109 AIR PARK RD S E ALBUQUERQUE NM 87106	1	NIST STRUCTURE & MECHANICS GRP POLYMER DIV POLYMERS RM A209
1	SAIC R ACEBAL 1225 JOHNSON FERRY RD STE 100 MARIETTA GA 30068	. 1	G MCKENNA GAITHERSBURG MD 20899 GENERAL DYNAMICS LAND
1	SAIC G CHRYSSOMALLIS 3800 W 80TH ST STE 1090	-	SYSTEM DIVISION D BARTLE PO BOX 1901 WARREN MI 48090
	BLOOMINGTON MN 55431	4	INSTITUTE FOR ADVANCED
6	ALLIANT TECHSYSTEMS INC C CANDLAND R BECKER L LEE R LONG		TECHNOLOGY H FAIR P SULILVAN W REINECKE I MCNAB
	D KAMDAR G KASSUELKE 600 2ND ST NE HOPKINS MN 55343-8367	1	4030 2 W BRAKER LN AUSTIN TX 78759  PM ADVANCED CONCEPTS LORAL VOUGHT SYSTEMS
1	CUSTOM ANALYTICAL ENGR SYS INC A ALEXANDER 13000 TENSOR LANE NE FLINTSTONE MD 21530		J TAYLOR MS WT 21 PO BOX 650003 DALLAS TX 76265-0003

NO. OF COPIES	ORGANIZATION	NO OF. COPIES	<u>ORGANIZATION</u>
2	UNITED DEFENSE LP P PARA G THOMASA 1107 COLEMAN AVE BOX 367 SAN JOSE CA 95103	4	NIST POLYMERS DIVISION R PARNAS J DUNKERS M VANLANDINGHAM D HUNSTON
1	MARINE CORPS SYSTEMS CMD PM GROUND WPNS COL R OWEN 2083 BARNETT AVE STE 315 QUANTICO VA 22134-5000	1	OAK RIDGE NATIONAL LAB A WERESZCZAK BLDG 4515 MS 6069
1	OFFICE OF NAVAL RES J KELLY 800 NORTH QUINCEY ST	1	PO BOX 2008 OAKRIDGE TN 37831-6064 COMMANDER
1	ARLINGTON VA 22217-5000 NAVSEE OJRI	-	USA ARDEC INDUSTRIAL ECOLOGY CTR T SACHAR
-	G CAMPONESCHI 2351 JEFFERSON DAVIS HWY ARLINGTON VA 22242-5160		BLDG 172 PICATINNY ARSENAL NJ 07806-5000
1	USAF WL MLS O L A HAKIM 5525 BAILEY LOOP 243E MCCLELLAN AFB CA 55552	1	COMMANDER USA ATCOM AVIATION APPLIED TECH DIR J SCHUCK FT EUSTIS VA 23604
1	NASA LANGLEY J MASTERS MS 389 HAMPTON VA 23662-5225	1	COMMANDER USA ARDEC AMSTA AR SRE
2	FAA TECH CTR D OPLINGER AAR 431 P SHYPRYKEVICH AAR 431 ATLANTIC CITY NJ 08405		D YEE PICATINNY ARSENAL NJ 07806-5000
1	NASA LANGLEY RC CC POE MS 188E NEWPORT NEWS VA 23608 USAF	1	COMMANDER USA ARDEC AMSTA AR QAC T D RIGOGLIOSO BLDG 354 M829E3 IPT PICATINNY ARSENAL NJ
1	WL MLBC E SHINN 2941 PST STE 1 WRIGHT PATTERSON AFB OH 45433-7750		07806-5000

		NO OF	
NO. OF		NO OF. COPIE <u>S</u>	ORGANIZATION
<u>COPIES</u>	<u>ORGANIZATION</u>	COPIES	ORGANIZATION
_	601 G 613 TO TO		AMSRL WM BC
7	COMMANDER		P PLOSTINS
	USA ARDEC		D LYON
	AMSTA AR CCH B		
	B KONRAD		J NEWILL
	E RIVERA		AMSRL WM BD
	G EUSTICE		S WILKERSON
	S PATEL		R FIFER
	G WAGNECZ		B FORCH
	R SAYER		R PESCE RODRIGUEZ
	F CHANG	,	B RICE
	BLDG 65		AMSRL WM
	PICATINNY ARSENAL NJ		D VIECHNICKI
	07806-5000		G HAGNAUER
			J MCCAULEY
6	DIRECTOR		AMSRL WM MA
	US ARMY RESEARCH LAB		R SHUFORD
	AMSRL WM MB		S MCKNIGHT
	A ABRAHAMIAN		L GHIORSE
	M BERMAN		AMSRL WM MB
	A FRYDMAN		V HARIK
	TLI		J SANDS
	W MCINTOSH		W DRYSDALE
	E SZYMANSKI		J BENDER
	2800 POWDER MILL RD		T BLANAS
	ADELPHI MD 20783-1197		T BOGETTI
	ADEDITION 20103 1191		R BOSSOLI
	ABERDEEN PROVING GROUND		L BURTON
	ADERDEENT ROVING GROONS	•	S CORNELISON
67	DIR USARL	•	P DEHMER
07	AMSRL CI		R DOOLEY
	AMSRL CI C		B FINK
	W STUREK		G GAZONAS
	AMSRL CI CB		S GHIORSE
	R KASTE		D GRANVILLE
	AMSRL CI S		D HOPKINS
	A MARK		C HOPPEL
	A MARK AMSRL SL B		D HENRY
	AMSRL SL B AMSRL SL BA		R KASTE
			M LEADORE
	AMSRL SL BE		R LIEB
	D BELY		E RIGAS
	AMSRL WM B		D SPAGNUOLO
	A HORST		W SPURGEON
	E SCHMIDT		J TZENG
	AMSRL WM BE		AMSRL WM MC
	G WREN		J BEATTY
	CLEVERITT		AMSRL WM MD
	D KOOKER		W ROY
			AMSRL WM T
			B BURNS

NO. OF

COPIES ORGANIZATION

#### ABERDEEN PROVING GROUND (CONT)

AMSRL WM TA

W GILLICH

E RAPACKI

T HAVEL

AMSRL WM TC

**R COATES** 

W DE ROSSET

AMSRL WM TD

W BRUCHEY

A D GUPTA AMSRL WM BB

**H ROGERS** 

AMSRL WM BA

F BRANDON

W D AMICO

AMSRL WM BR

J BORNSTEIN

AMSRL WM TE

A NIILER

AMSRL WM BF

**JLACETERA** 

INTENTIONALLY LEFT BLANK.

#### Form Approved REPORT DOCUMENTATION PAGE OMB No. 0704-0188 n for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sourceding the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this gentering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Projection 74-0183), Washington, DC 20503. 1. AGENCY USE ONLY (Leave blank) 2. REPORT DATE 2. REPORT Type AND DATE OF The AND DATE 1. AGENCY USE ONLY (Leave blank) Final, January 1997 - January 1998 December 1999 4. TITLE AND SUBTITLE 5. FUNDING NUMBERS AH42 The Role of Transverse Flow in Co-Injection Resin Transfer Molding 6. AUTHOR(S) Bruce K. Fink, Emanuele F. Gillio,\* Surresh G. Advani,\* and John W. Gillespie Jr.\* 8. PERFORMING ORGANIZATION 7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) REPORT NUMBER U.S. Army Research Laboratory ARL-TR-2135 ATTN: AMSRL-WM-MB Aberdeen Proving Ground, MD 21005-5069 10.SPONSORING/MONITORING 9. SPONSORING/MONITORING AGENCY NAMES(S) AND ADDRESS(ES) AGENCY REPORT NUMBER 11. SUPPLEMENTARY NOTES \*University of Delaware, Newark, DE 19716 12b. DISTRIBUTION CODE 12a. DISTRIBUTION/AVAILABILITY STATEMENT Approved for public release; distribution is unlimited. 13. ABSTRACT (Maximum 200 words) A co-injection resin transfer molding (CIRTM) process has been developed at the U.S. Army Research Laboratory (ARL) in collaboration with the University of Delaware. It enables two or more resins to be simultaneously injected into a mold filled with a stationary fiber preform. This process allows for the manufacturing of cocured multilayer multiresin structures in a single processing step. A separation layer is used to provide resin compatibility during cure and to control resin mixing. In this study, scaling issues relating the role of transverse permeability in resin mixing are investigated. This report presents two different approaches taken to understand the causes of transverse flow and to quantify the amount of transverse flow. The first approach is a one-dimensional (1-D) model, which explains the important parameters that govern the flow in CIRTM. The second approach is based on an existing finite element code, which is modified to allow for the injection of multiple resins. The total amount of transverse flow was quantified using the finite element code. This research shows that he CIRTM process requires a totally impermeable separation layer if CIRTM is used to manufacture large parts and/or if the resins injected have significantly different viscosities. 15. NUMBER OF PAGES 14. SUBJECT TERMS co-injection resin transfer molding, vinyl ester, phenolic, flammability, composite materials 16. PRICE CODE 20. LIMITATION OF ABSTRACT 17. SECURITY CLASSIFICATION 18. SECURITY CLASSIFICATION 19. SECURITY CLASSIFICATION

SAR

UNCLASSIFIED

OF REPORT

OF THIS PAGE

UNCLASSIFIED

OF ABSTRACT

UNCLASSIFIED

INTENTIONALLY LEFT BLANK.

#### USER EVALUATION SHEET/CHANGE OF ADDRESS

This Laboratory undertakes a continuing effort to improve the quality of the reports it publishes. Your comments/answers to the items/questions below will aid us in our efforts.

1. ARL Report Num	ber/Author ARL-TR-2135 (Fink)	Date of Report December 1999
2. Date Report Rece	ived	·
· ·	atisfy a need? (Comment on purpose, r	related project, or other area of interest for which the report will
		source, design data, procedure, source of ideas, etc.)
avoided, or efficienc	ies achieved, etc? If so, please elabora	e savings as far as man-hours or dollars saved, operating costs ite.
technical content, for	rmat, etc.)	ed to improve future reports? (Indicate changes to organization,
	Organization	
CURRENT	Name	E-mail Name
ADDRESS	Street or P.O. Box No.	
	City, State, Zip Code	
7. If indicating a Cha or Incorrect address		please provide the Current or Correct address above and the Old
	Organization	
OLD	Name	
ADDRESS	Street or P.O. Box No.	· · · · · · · · · · · · · · · · · · ·
	City, State, Zip Code	
	(Remove this sheet, fold as i	ndicated, tape closed, and mail.)

(Remove this sheet, fold as indicated, tape closed, and mail.)
(DO NOT STAPLE)